# NATIONAL BUREAU OF STANDARDS REPORT

8359

PROPERTY UP SUUTINEEST RESEARCH INSTITUTE LIBRANT SAN ANTONIO, TEXAS

# EVALUATION OF RESISTANCE STRAIN GAGES AT ELEVATED TEMPERATURES

Progress Report No. 16

by

J. T. Trumbo and R. L. Bloss



U. S. DEPARTMENT OF COMMERCE NATIONAL BUREAU OF STANDARDS

# THE NATIONAL BUREAU OF STANDARDS

The National Bureau of Standards is a principal focal point in the Federal Government for assuring maximum application of the physical and engineering sciences to the advancement of technology in industry and commerce. Its responsibilities include development and maintenance of the national standards of measurement, and the provisions of means for making measurements consistent with those standards; determination of physical constants and properties of materials; development of methods for testing materials, mechanisms, and structures, and making such tests as may be necessary, particularly for government agencies; cooperation in the establishment of standard practices for incorporation in codes and specifications; advisory service to government agencies on scientific and technical problems; invention and development of devices to serve special needs of the Government; assistance to industry, business, and consumers in the development and acceptance of commercial standards and simplified trade practice recommendations; administration of programs in cooperation with United States business groups and standards organizations for the development of international standards of practice; and maintenance of a clearinghouse for the collection and dissemination of scientific, technical, and engineering information. The scope of the Bureau's activities is suggested in the following listing of its four Institutes and their organizational units.

Institute for Basic Standards. Electricity. Metrology. Heat. Radiation Physics. Mechanics. Applied Mathematics. Atomic Physics. Physical Chemistry. Laboratory Astrophysics.\* Radio Standards Laboratory: Radio Standards Physics; Radio Standards Engineering.\*\* Office of Standard Reference Data.

Institute for Materials Research. Analytical Chemistry. Polymers. Metallurgy. Inorganic Materials. Reactor Radiations. Cryogenics.\*\* Office of Standard Reference Materials.

Central Radio Propagation Laboratory.\*\* Ionosphere Research and Propagation. Troposphere and Space Telecommunications. Radio Systems. Upper Atmosphere and Space Physics.

Institute for Applied Technology. Textiles and Apparel Technology Center. Building Research. Industrial Equipment. Information Technology. Performance Test Development. Instrumentation. Transport Systems. Office of Technical Services. Office of Weights and Measures. Office of Engineering Standards. Office of Industrial Services.

\*\* Located at Boulder, Colorado.

<sup>\*</sup> NBS Group, Joint Institute for Laboratory Astrophysics at the University of Colorado.

# NATIONAL BUREAU OF STANDARDS REPORT

NBS PROJECT NBS REPORT

0604-20-06441

May 1964

8359

# EVALUATION OF RESISTANCE STRAIN GAGES AT ELEVATED TEMPERATURES

Progress Report No. 16

bу

J. T. Trumbo and R. L. Bloss

Engineering Mechanics Section Division of Mechanics

Technical Report

to

Bureau of Naval Weapons Aeronautical Systems Division

Order No. IPR-19-64-8022-WEPS

#### IMPORTANT NOTICE

NATIONAL BUREAU OF STA for use within the Government. E and review. For this reason, the whole or in part, is not authoriz Bureau of Standards, Washingtor the Report has been specifically p Approved for public release by the Director of the National Institute of Standards and Technology (NIST) on October 9, 2015.

accounting documents intended ubjected to additional evaluation listing of this Report, either in Office of the Director, National he Government agency for which pies for its own use.



# U. S. DEPARTMENT OF COMMERCE NATIONAL BUREAU OF STANDARDS

#### FOREWORD

In recent years the use of structures at elevated temperatures has increased greatly. If the safe design and efficient use of structural materials are to be assured, a knowledge of the mechanical properties of materials and of structural configurations is essential. In determining these properties, the measurement of strains and deformations is important. Strain gages to measure these quantities must be capable of operating satisfactorily over a wide temperature range.

In order to determine the characteristics of strain gages that are available for use at elevated temperatures, the Department of the Navy and the Department of the Air Force have sponsored a program for the evaluation of these gages. Results obtained from only one gage type are given in this report so that performance information may be made available without undue delay. Results obtained from other gage types have been presented in earlier reports of this series.

There is a continuing effort on the part of manufacturers and research organizations to develop improved strain gages for use at elevated temperatures. Therefore the results given in this report would not necessarily show the performance of similar gages which may differ in characteristics due to differences in materials, treatments, or methods of fabrication.

L. K. Irwin
Chief, Engineering Mechanics
Section

B. L. Wilson Chief, Mechanics Division

# CONTENTS

		Page
FOR	WORD	II
SYN	PSIS	1
1.	INTRODUCTION	1
2.	DESCRIPTION OF GAGES	2
3.	TEST EQUIPMENT AND METHODS	2
4.	RESULTS AND DISCUSSION	2
	4.1 Strain Sensitivity	2
	4.2 Variation of Gage Factor with Temperature	3
	4.3 Large Strains	3
	4.4 Drift	4
	4.5 Temperature Sensitivity	4
	4.6 Transient Heating	4
	4.7 Leakage Resistance	5
5.	CONCLUSIONS	5
6.	MANUFACTURER'S DATA	6
7	REPERFNERS	0

# EVALUATION OF RESISTANCE STRAIN GAGES AT ELEVATED TEMPERATURES

Progress Report No. 16

bу

J. T. Trumbo and R. L. Bloss

## Synopsis

Type FNW-FB-9-50-12 resistance strain gages, manufactured by the Baldwin-Lima-Hamilton Corporation, were evaluated. The results of these tests indicate that the gage factor for compressive loading is somewhat higher than for tensile loading; that the gage factor decreases with increasing temperature by about one-half percent per  $100^{\circ}$  F; that large errors can be expected when strains greater than 0.002 are measured; that the gages are well compensated for resistance instability; that the temperature sensitivity is low and repeatable from gage to gage, but the gage response is strongly affected by heating rate; and that the leakage resistance is influenced by temperature and the thermal history of the gage.

#### 1. INTRODUCTION

In the continuing evaluation of resistance strain gages designed for use at elevated temperatures, type FNW-FB-9-50-12 gages of the Baldwin-Lima-Hamilton Corporation were tested to determine the following characteristics:

- (1) Gage factor at room temperature,
- (2) Variation of gage factor with temperature,
- (3) Behavior when subjected to large strains,
- (4) Change in indicated strain with time at constant temperature (drift),
- (5) Change in indicated strain due to temperature changes,
- (6) Behavior under transient heating conditions, and
- (7) Resistance between the gage and the material to which it is attached.

The results of previous evaluations of other types of gages are given in references 1 through 13.

#### 2. DESCRIPTION OF GAGES

Each of the gages consisted of four resistive elements as shown in figure 1. These elements are bonded to a piece of 0.005 inch thick shim stock by the manufacturer. For use, a gage is attached to a metallic surface by two rows of spotwelds along the center line of the shim material, and the gage leads are connected to form a full-bridge circuit (figure 1). A low temperature sensitivity over a large temperature range is sought by the proper choice of shim material and appropriate manufacturing techniques. According to the manufacturer, the gages tested were prestabilized, post cured and ready for use in the temperature range of  $-400^{\circ}$  to  $1000^{\circ}$  F. The gages were ordered for use on material having a linear temperature coefficient of expansion of  $9x10^{-6}$  per degree F.

# 3. TEST EQUIPMENT AND METHODS

The equipment and methods used for all these evaluation tests have been described in references 5, 8, 14, 15, and 16.

# 4. RESULTS AND DISCUSSION

The number of gages subjected to the various tests and the voltages applied to the gage circuits are shown in table 1. The heating rates used for the transient heating test are given in table 2. The results of the evaluation tests are given in table 3 and figures 2 through 24.

In order that the results of the tests might be readily compared with previous reports of this series, the gage response is given in terms of the relative change of resistance of one arm of a bridge circuit that would produce the same electrical output that was obtained from the full bridge circuit of the gage.

#### 4.1 Strain Sensitivity

Gage factor values were obtained at about 75° F from four gages at a maximum strain of about 0.001 in both tension and compression. The gage response was determined by comparing the electrical output with the output of another bridge circuit of which two arms were a Wenner ratio set. Power to the bridge circuits was from separate power supplies, but the input voltages were compared and adjusted to be equal just prior to each reading. Readings were taken at predetermined settings of the Wenner ratio set by straining the specimen to which the gage was attached until the outputs of the bridge circuits were equal. The actual strain to which the gage was subjected was determined with a Tuckerman extensometer.

The results of these tests are given in table 3 where

 $K_{ij}$  = the gage factor for increasing load,

 $K_{d}$  = the gage factor for decreasing load,

 $\tilde{K}$  = the average gage factor.

Gages  ${\rm A}_1$  and  ${\rm A}_3$  were tested in tension before being tested in compression. Gages  ${\rm A}_2$  and  ${\rm A}_4$  were tested in compression first.

The differences between the experimental gage factor values and the manufacturer's nominal value, 3.14, are shown in figure 2. The departure of a plotted point from the origin shows the difference between the experimental value and the nominal value. The departure of the points from the diagonal line show the differences between gage factor values for tension and compression loading. Results show that the gage factor values for compression loading were higher than for tension loading and that the value for the first loading cycle differed from values for subsequent tests.

Figures 3 through 6 show the departure from linearity of the gage response and the zero shift for the first and third loading cycles. The maximum strain was about 0.001. The gage factor, K, used in the data reduction was the manufacturer's nominal value, 3.14. The arrows on the curves indicate the direction of loading. No corrections were applied for temperature fluctuations during the tests.

# 4.2 Variation of Gage Factor with Temperature

The variations of gage factor with increasing temperature, obtained by dynamic test methods, are shown in figures 7 through 9. Each curve represents the change of gage factor of one gage during one test. Each figure gives the results obtained from one gage to show the repeatability of the gages from test to test.

Figure 10 shows the average variation of gage factor of three gages for each of the test runs and the extreme values obtained for any gage during any test. This shows that the gage factor tended to decrease in a linear fashion to  $1000^\circ$  F. The gage factors at  $1000^\circ$  F were about five percent less than at room temperature.

# 4.3 Large Strains

The results of tests in which the gages were subjected to tessile strains greater than those used for gage factor determinations are shown in figures 11 and 12. In order to compute the strains indicated by the resistance gage,  $\epsilon_{\mbox{Ind}} = \frac{1}{K} \; \frac{\triangle \! R}{R} \; , \; \; \mbox{at room temperature the gage factor value, K, used was the}$ 

 $\epsilon_{\rm Ind} = \frac{1}{K} = \frac{1}{R}$ , at room temperature the gage factor value, K, used was the average of all values obtained from the room temperature gage factor tests in tension. For the large strain test at 900° F, the room temperature value was adjusted by the average amount of change found during the first test runs of the variation of gage factor with temperature tests.

The errors of these gages tended to increase rapidly with increasing strain, exceeding ten percent of the indicated strain for strains greater than about 0.002 for all tests at room temperature and 900° F. The shape of the curves would indicate that the strain is not being transmitted to the gage properly or that the compensating arms of the gage are being strained a proportionately greater amount as the strain increases. Tests were discontinued at strains of about 0.003.

## 4.4 Drift

Records of the change of gage output with time for three gages at constant test temperatures up to  $1200^{\circ}$  F are shown in figures 13 through 19. The results were obtained after heating the gage installation at about  $10^{\circ}$  F per second from room temperature or the next lower test temperature. Recording was started one minute after the desired test temperature was reached. The second series of tests, Run 2, were made after the gages had been tested once at each test temperature up to  $1000^{\circ}$  F. The temperature fluctuations during the tests exceeded  $3^{\circ}$  F during only one test of one gage during the second test series at  $800^{\circ}$  F. The data was not corrected for temperature fluctuations.

The results indicate that the gages are well compensated for resistance instability effects which did not exceed  $2 \times 10^{-3}$  for 30 minutes to  $1200^{\circ}$  F. At temperatures up to  $1000^{\circ}$  F, the greatest average drift rate for thirty minutes was less than  $10^{-5}$  per minute (apparent relative resistance change of one gage arm). This is equivalent to an apparent strain of about  $3 \times 10^{-6}$  per minute.

## 4.5 Temperature Sensitivity

Average values of the change of gage output with increasing temperature for three gages are shown in figures 20 through 22. The maximum and minimum values obtained during the tests are also shown. During these tests, the gages were heated at about 10° F per second. Tests 1 and 2 were carried to a maximum temperature of about 1000° F. Tests 3, 4 and 5 were carried to a maximum temperature of about 1600° F. Results are shown for tests 1, 3 and 5 only since the values for tests 2 and 3 were in good agreement at temperatures up to 1000° F, and the values for tests 4 and 5 were in good agreement for temperatures up to 1600° F. Each point on the graphs was determined as the slope of a line drawn tangent to a curve of gage output versus temperature. Results for a portion of the fifth test were lost when the x-y recorder went off scale.

## 4.6 Transient Heating

The results of tests in which the temperature of the test strip to which the gage was attached was increased at about  $2^{\circ}$ ,  $10^{\circ}$ ,  $25^{\circ}$ ,  $50^{\circ}$  and  $80^{\circ}$  F per second are shown in figure 23. The results of all tests are not shown because the differences between tests at the same heating rate were not thought to be

significant after the first test. Tests were also made with no power to the gage circuit to determine the possible effect of uncompensated thermal emfs within the circuit. Results of these tests indicated that the effects, if any, are small.

The results shown in figure 23 indicate that these gages are quite sensitive to heating rate, at least when radiant heating techniques are used. This is probably due to the poor thermal contact between the gage shim material and the test strip at the outer edge where the compensating arms of the gage are located and the intimate contact created by the spot welds along the center line near the active arms. This difference in contact with a heat sink (or source) and the low thermal conductivity of the shim material could produce large thermal gradients in the gage when high heating rates are encountered.

## 4.7 Leakage Resistance

Typical results of tests to determine the resistance between the gage and the test strip as a function of temperature are shown in figure 24. The results were recorded while the test strip was being heated at a rate of about 10° F per second to the maximum test temperature. Two tests to a maximum temperature of about 1000° F were followed by three tests to about 1600° F. Four gages were tested, but, since the results were nearly the same for all gages, the results for only one gage are shown. These results show that, even though these gages have been post cured during their manufacture, there is considerable improvement in the leakage resistance after the first heating cycle to 1000° F. Further improvement is found after heating the gages to 1600° F. The values shown can be considered to be only a qualitative indication of the insulating properties of the cement since it has been found that ceramic cements do not follow Ohm's law (reference 17). The small negative leakage resistance shown at higher temperatures is probably due to an emf being generated between the test strip material and the gage element.

## 5. CONCLUSIONS

For gages of this type, the data obtained from the evaluation tests covered by this report indicate that:

- (1) Gage factor values for compression loading were, on the average, about five percent higher than for tension loading. For both modes of loading, the gage factor varied as much as  $\pm$  10 percent from the manufacturer's value.
- (2) The gage factor decreases in a linear manner with increasing temperature to 1000° F. At 1000° F the value is about five percent below the value at room temperature.
- (3) At strains of 0.002, errors of indicated strain exceeded ten percent at room temperature and at 900° F.

- (4) The gages showed an average drift of less than  $3 \times 10^{-6}$  inches per inch per minute for 30 minutes at temperatures up to  $1000^{\circ}$  F. This low value shows good drift compensation by this gage configuration.
- (5) The temperature sensitivity did not exceed 11 parts per million resistance change per degree Fahrenheit to  $1000^{\circ}$  F.
- (6) The gages were sensitive to heating rate, at least when radiant heating methods were used.
- (7) The resistance between the gage and the test strip depends upon the temperature and the thermal history of the gage. This resistance was greatly increased by heating to 1000° F or higher.

# 6. MANUFACTURER'S DATA

Correspondence with the manufacturer (reference 18) has revealed that data gathered in their laboratory do not fully agree with the information contained in this report. In particular, their data show:

- (1) A gage factor of  $3.28 \pm 3$  percent at strains of 0.001
- (2) The compression gage factor being higher than the tension factor by the following amounts:
  - (a) 1.3 to 3.6 percent at 0.001 strain
  - (b) 0.6 to 2.4 percent at 0.002-strain
  - (c) 1.3 percent at 0.003 strain
- (3) Maximum error of indicated strain of 2.7 percent at 0.002 strain and 4.7 percent at 0.003 strain.
- (4) Zero shifts not greater than one percent for second and third load cycles at any strain level.

It was also indicated that the manufacturer has found that the installation of the gage is somewhat critical. The installation instructions are said to have been revised to provide the user with better instructions than were previously available. Improvements in manufacturing techniques since the initial release of these gages are also claimed.

As noted in the foreword of this report, and previous reports of this series, strain measurement at elevated temperatures is not a static field. Improvements in materials, methods of fabrication, and techniques in handling the strain gages may improve gage characteristics significantly. The manufacturers of these gages are continually trying to make such improvements to provide the user with better gages.

The authors gratefully acknowledge the assistance of T. W. Butler, C. H. Melton, M. L. Sundquist, and R. J. Wall in performing the evaluation tests and in preparing this report.

#### 7. REFERENCES

- (1) R. L. Bloss and C. H. Melton, "An Evaluation of Two Types of Resistance Strain Gages at Temperatures up to 600° F," NBS Report No. 4676, May 1956 (ASTIA No. AD 94696).
- (2) R. L. Bloss and C. H. Melton, "An Evaluation of One Type of Resistance Strain Gage at Temperatures up to 600° F," NBS Report No. 4747, July 1956 (ASTIA No. AD 101079).
- (3) R. L. Bloss and C. H. Melton, "An Evaluation of Two Types of Resistance Strain Gages at Temperatures up to 600° F," NBS Report No. 4843, September 1956 (ASTIA No. AD 107662).
- (4) R. L. Bloss and C. H. Melton, "An Evaluation of Strain Gages Designed for Use at Elevated Temperatures -- Preliminary Tests for Temperatures up to 1000° F," NBS Report No. 5286, May 1957 (ASTIA No. AD 135050).
- (5) R. L. Bloss and C. H. Melton, "Evaluation of Resistance Strain Gages at Elevated Temperatures" (Progress Report No. 5), NBS Report No. 6117, August 1958 (ASTIA No. AD 202419L).
- (6) R. L. Bloss, C. H. Melton, and M. L. Seman, "Evaluation of Resistance Strain Gages at Elevated Temperatures" (Progress Report No. 6), NBS Report No. 6245, December 1958, (ASTIA No. AD 211391).
- (7) R. L. Bloss, C. H. Melton, and M. L. Seman, "Evaluation of Resistance Strain Gages at Elevated Temperatures" (Progress Report No. 7) NBS Report No. 6395, April 1959, (ASTIA No. AD 217651).
- (8) R. L. Bloss, C. H. Melton, and J. T. Trumbo, "Evaluation of Resistance Strain Gages at Elevated Temperatures" (Frogress Report No. 8) NBS Report No. 6526, August 1959, (ASTIA No. AD 227197).
- (9) R. L. Bloss, C. H. Melton, and J. T. Trumbo, "Evaluation of Resistance Strain Gages at Elevated Temperatures" (Progress Report No. 9) NBS Report No. 6900, July 1960, (ASTIA No. AD 240829).
- (10) R. L. Bloss, J. T. Trumbo, and G. H. Melton, "Evaluation of Resistance Strain Gages at Elevated Temperatures" (Progress Report No. 10) NBS Report No. 7003, November 1960, (ASTIA No. AD 262059).
- (11) J. T. Trumbo, C. H. Melton, and R. L. Bloss, "Evaluation of Resistance Strain Gages at Elevated Temperatures" (Progress Report No. 12) NBS Report No. 7161, May 1961 (ASTIA No. AD 262790).

- (12) R. L. Bloss, J. T. Trumbo, C. H. Melton and J. S. Steel "Evaluation of Resistance Strain Gages at Elevated Temperatures" (Progress Report No. 13) NBS Report No. 7399, December 1961 (ASTIA No. AD 281606).
- (13) R. L. Bloss, J. T. Trumbo, C. H. Melton and J. S. Steel "Evaluation of Resistance Strain Gages at Elevated Temperatures" (Progress Report No. 14) NBS Report No. 7588, August 1962 (ASTIA No. AD 288224).
- (14) R. L. Bloss, "A Facility for the Evaluation of Resistance Strain Gages at Elevated Temperatures, Symposium on Elevated Temperature Strain Gages," ASTM Special Technical Publication No. 230, pp. 57-66.
- (15) R. L. Bloss, "Evaluation of Resistance Strain Gages at Elevated Temperatures," Materials Research and Standards, Vol. 1, No. 1, p. 9 (1961).
- (16) R. L. Bloss and J. T. Trumbo, "A Method for Measuring the Instability of Resistance Strain Gages at Elevated Temperatures," ISA Transactions, Vol. 2, No. 2 p. 112 (1963).
- (17) J. W. Pitts and D. G. Moore, "Development of High-Temperature Strain Gages" NBS Monograph 26, 1961.
- (18) Letter from L. J. Weymonth, Baldwin-Lima-Hamilton Corp., dated June 19, 1964.

Table 1 - Number of Gages Tested and Gage Circuit Voltage

	Number of	Voltage applied	
Type of test	gages tested	to gage Circuit	
		volts, d-c	
Gage factor determination	4	6	
Gage factor variation	4	6	
Large strain	4	3*	
Resistance instability (drift)	3	8	
Temperature sensitivity	3 .	8	
Transient heating	4	8	
Leakage resistance	4	10**	

<sup>\*</sup> a-c (1000 cps)

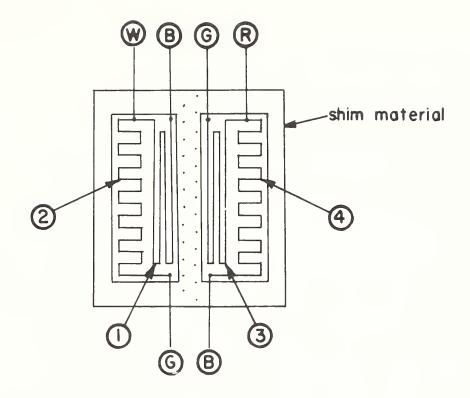
<sup>\*\*</sup> Maximum voltage between gage and test strip

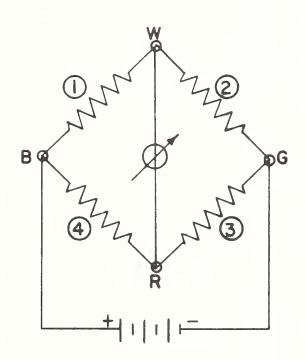
Table 2 - Heating Rates for Transient Heating Tests

Test No.	Nominal heating rate	Power to gage circuit	
	°F/sec		
l through 3	10	yes	
through 6	2	yes	
7	10	yes	
3 through 10	25	yes	
11	10	yes	
12 through 14	50	yes	
15	10	yes	
l6 through 18	80	yes	
19	10	yes	
20	10	no	
21	50	no	
22	80	no	
23	10	no	
24	50	no	
25	80	no	

Table 3 - Gage Factor Values at Room Temperature (Approximately 75° F)

				Gage fact	or values		
Gage	Run		Tension			Compression	
No.	No.	K <sub>u</sub>	K <sub>d</sub>	Ř.	K u	Kd	Ķ ———
A <sub>1</sub>	1 2 3	2.996 3.169 3.149	3.035 3.044 3.061	3.016 3.106 3.105	3.026 3.238 3.283	3.420 3.302 3.245	3.223 3.270 3.264
	Average			3.076			3.252
A <sub>2</sub>	1 2 3	3.132 3.356 3.374	3.345 3.327 3.330	3.238 3.342 3.352	3.384 3.525 3.440	3.506 3.457 3.588	3.445 3.491 3.514
	Average			3.311			3.483
<sup>A</sup> 3	1 2 3	2.742 3.106 2.959	3.040 3.035 3.058	2.891 3.070 3.008	2.820 3.035 3.039	3.188 3.116 3.193	3.004 3.076 3.116
	Average			2.990			3.065
A <sub>4</sub>	1 2 3	3.000 3.289 3.243	3.223 3.210 3.294	3.112 3.250 3.268	3.156 3.330 3.336	3.445 3.436 3.437	3.300 3.383 3.386
	Average			3.210			3.356





Suggested Wiring Color Code

B-Black

R-Red

G-Green

W-White

Fig. 1 Gage Configuration and Electrical Circuit

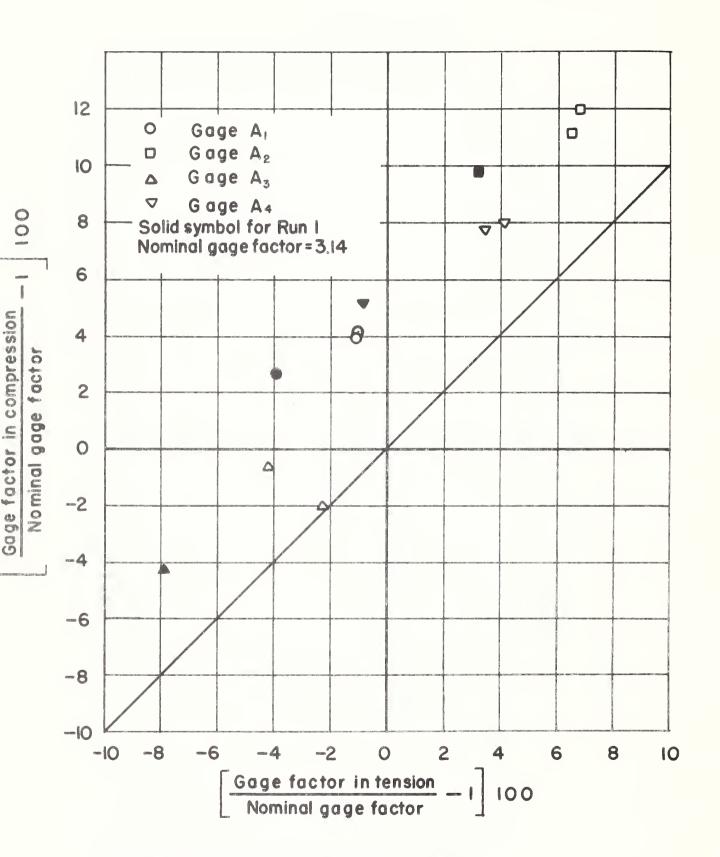


Fig.2 Gage factor deviation at 75° F

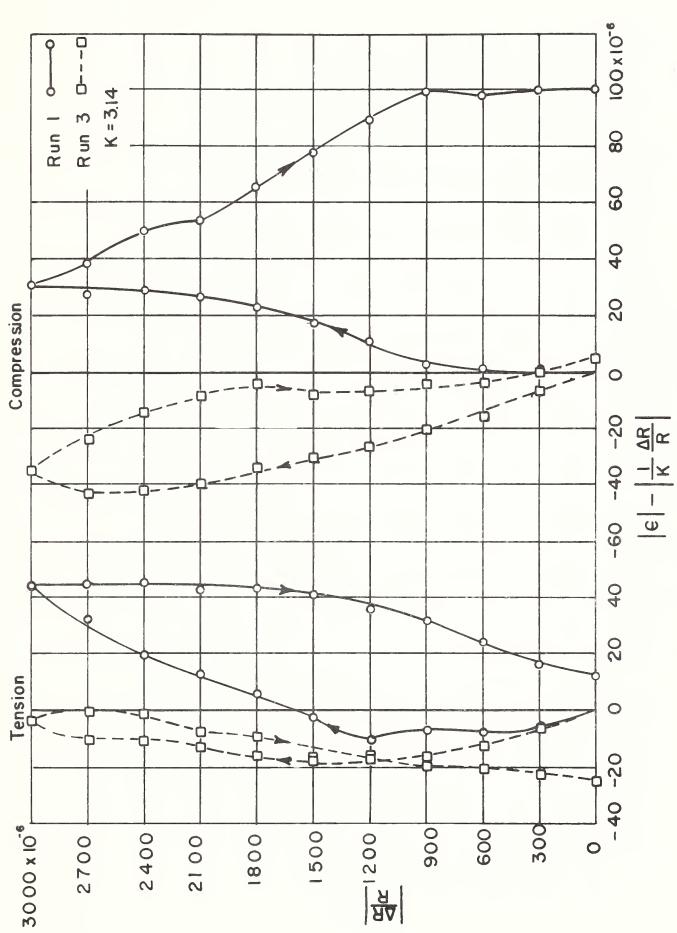
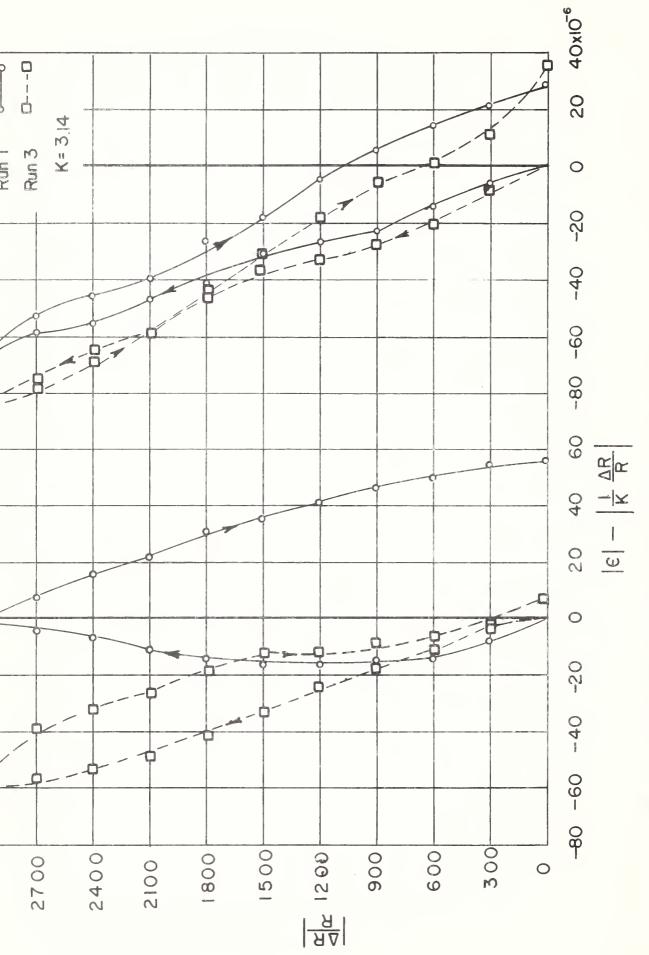
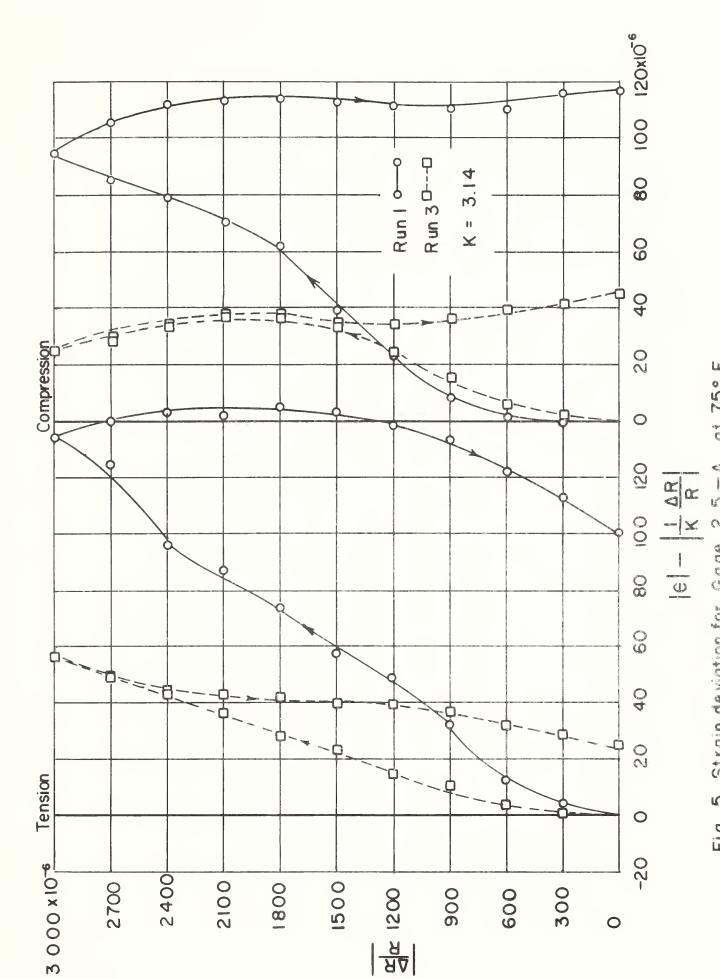
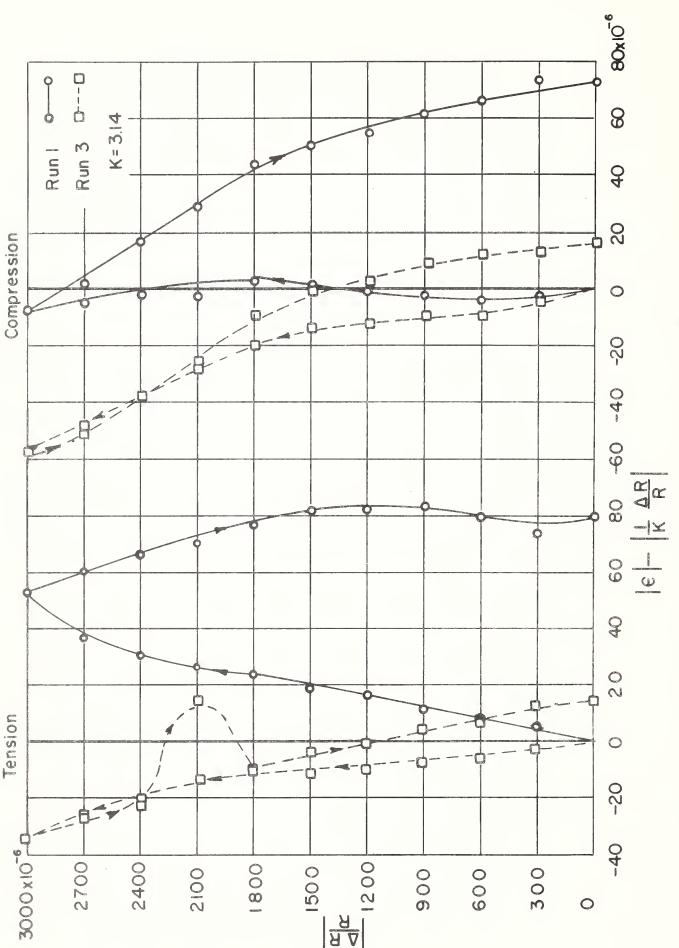


Fig. 3 Strain deviation for gage 2.5-A, at 75°F



Strain deviation for Gage 2.5-A<sub>2</sub> at 75° F Fig. 4





Strain deviation for Gage 2.5-A, at 75°F Fig. 6

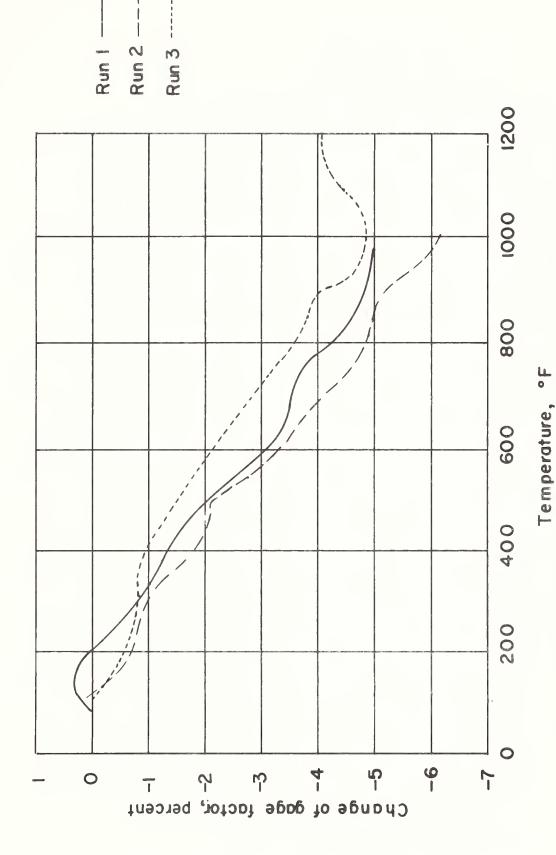


Fig.7 Variation of gage factor with temperature for Gage 2.5-B<sub>2</sub>

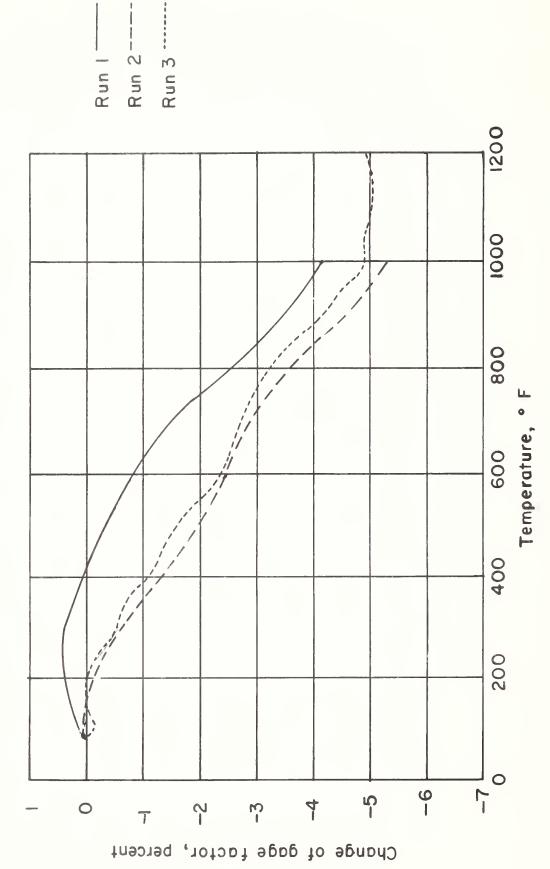


Fig. 8 Variation of gage factor with temperature for Gage  $2.5-B_3$ 

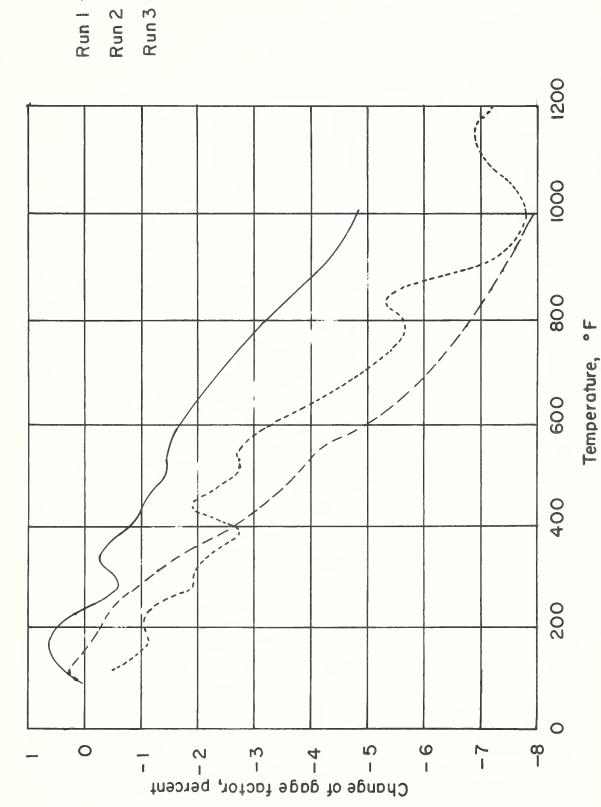


Fig. 9 Variation of gage factor with temperature for Gage 2.5-B.

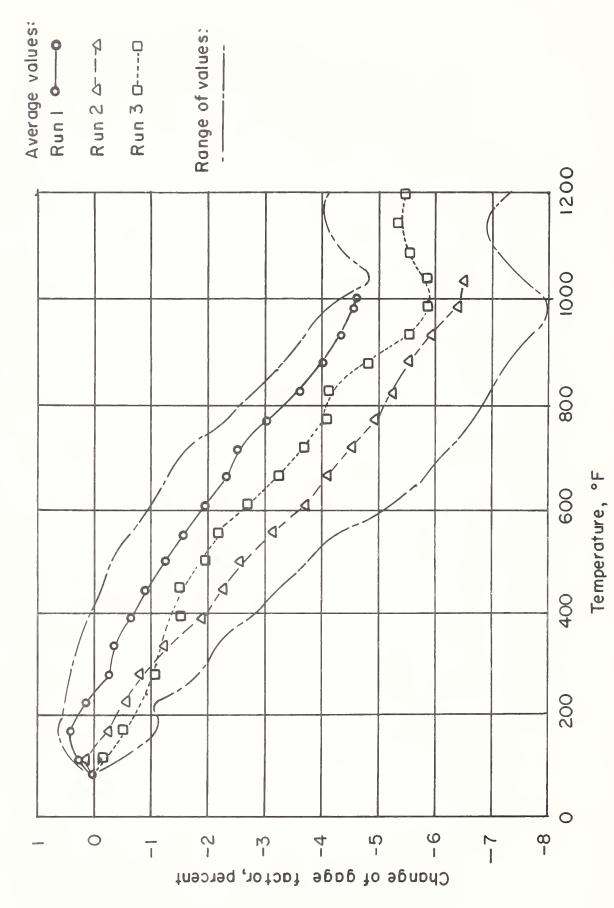
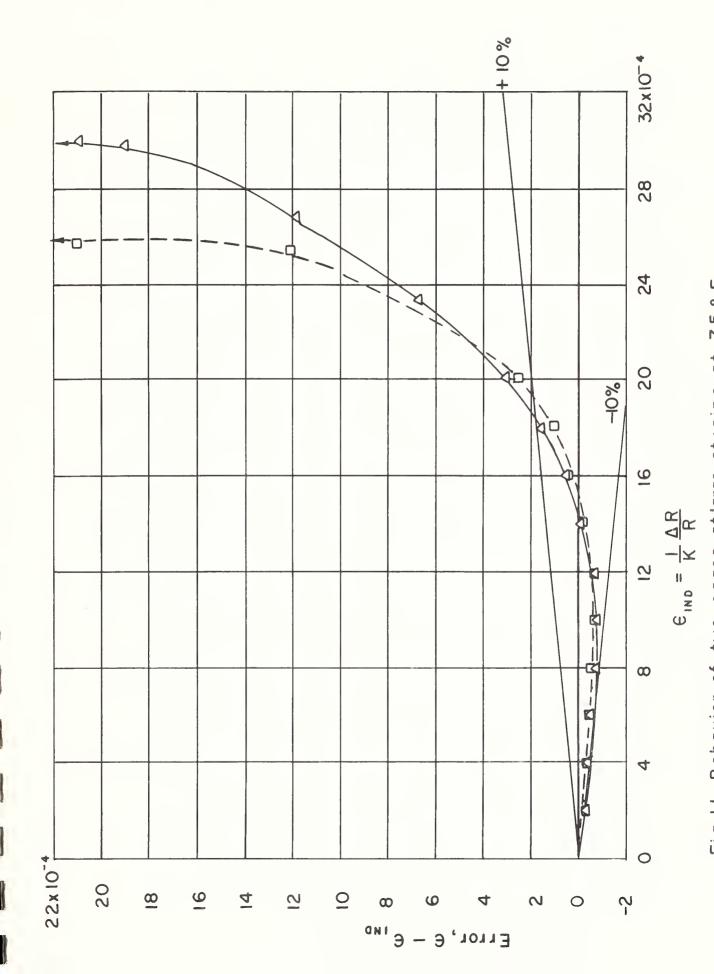
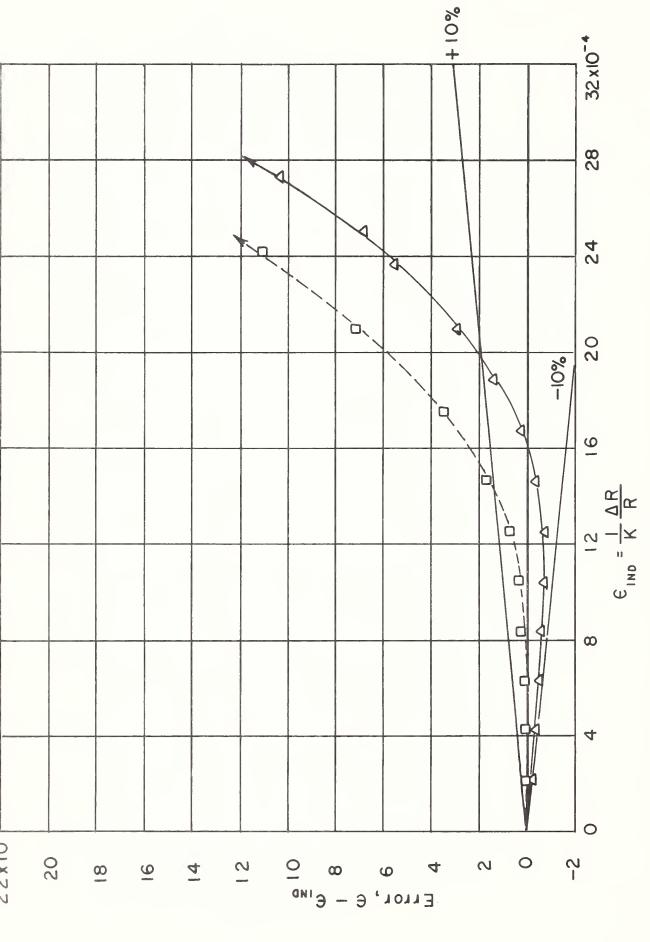
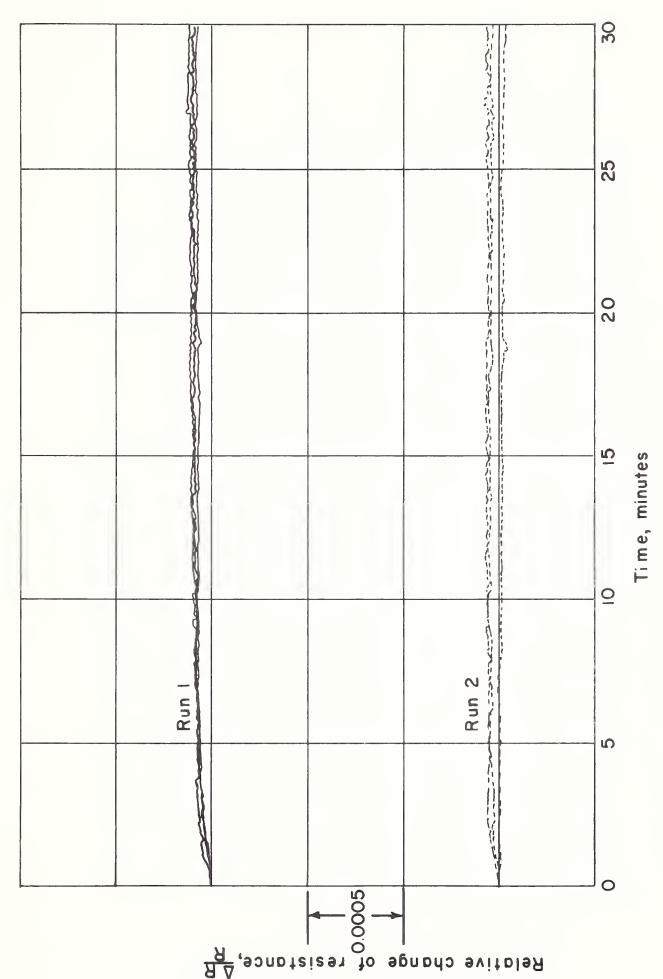


Fig.10 Variation of gage factor with temperature for three gages





Behavior of two gages at large strains at 900° F Fig. 12



Drift behavior of three gages at 600°F Fig. 13

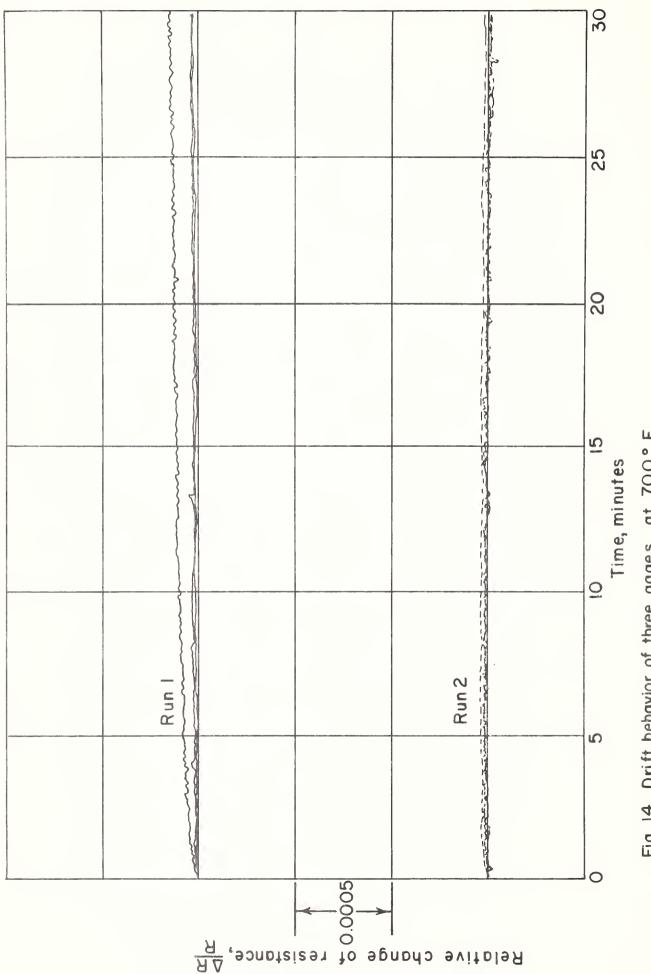


Fig. 14 Drift behavior of three gages at 700°F

Fig. 15 Drift behavior of three gages at 800°F

at 900 ° F Fig. 16 Drift behavior of three gages

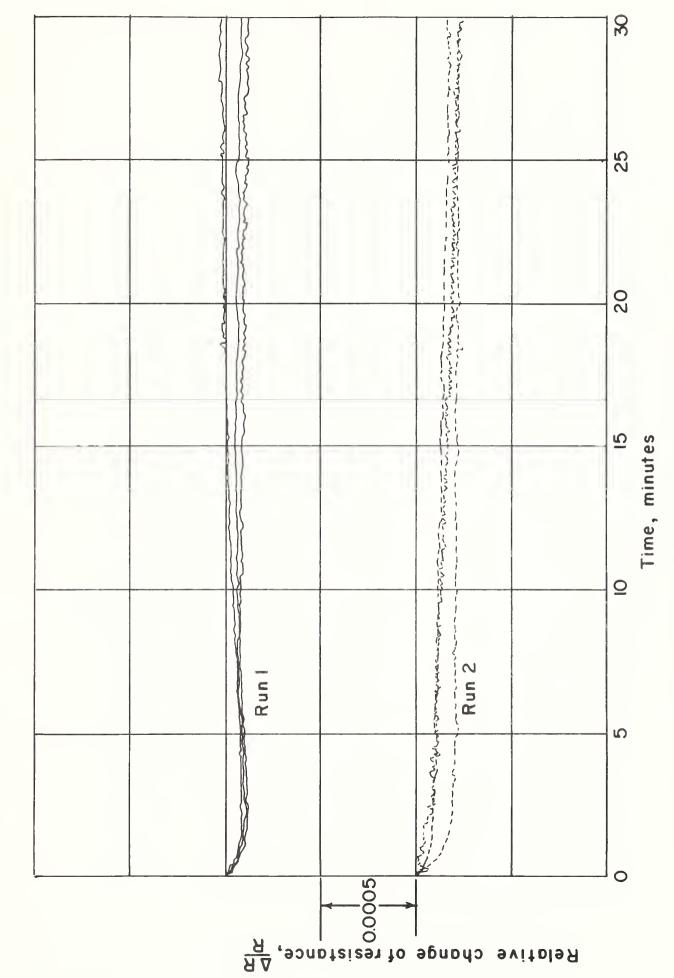
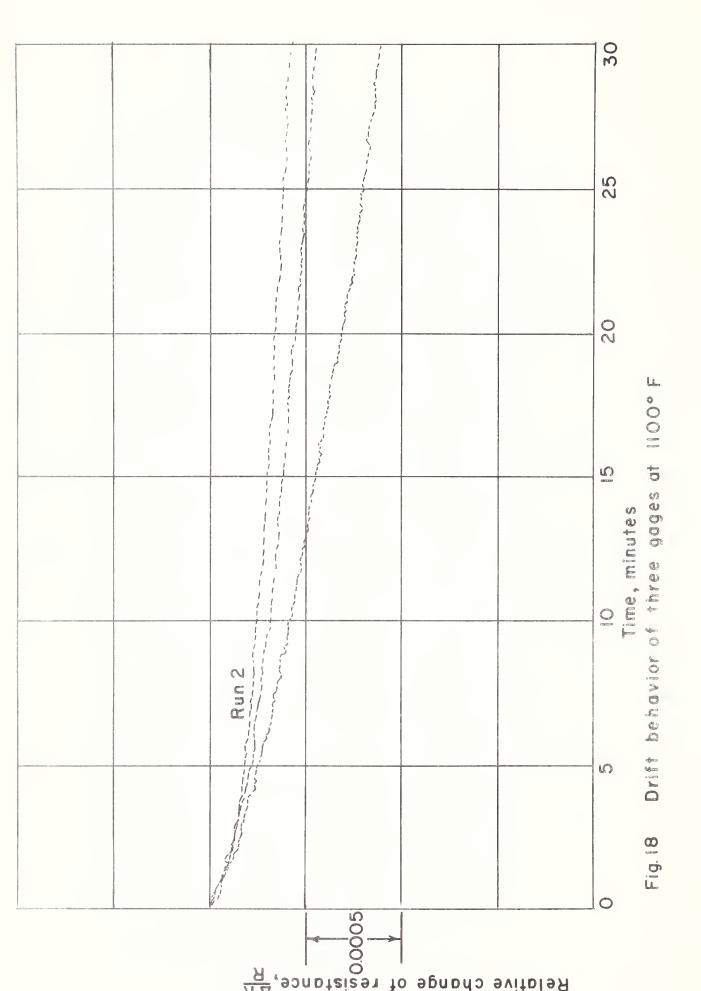


Fig. 17 Drift behavior of three gages at 1000°F



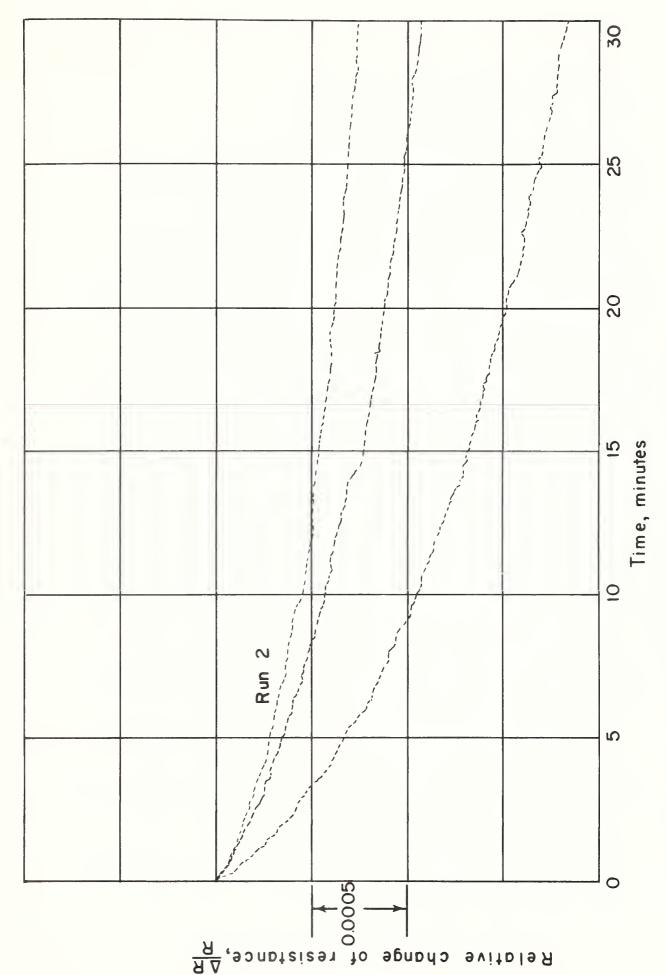


Fig.19 Drift behavior of three gages at 1200° F

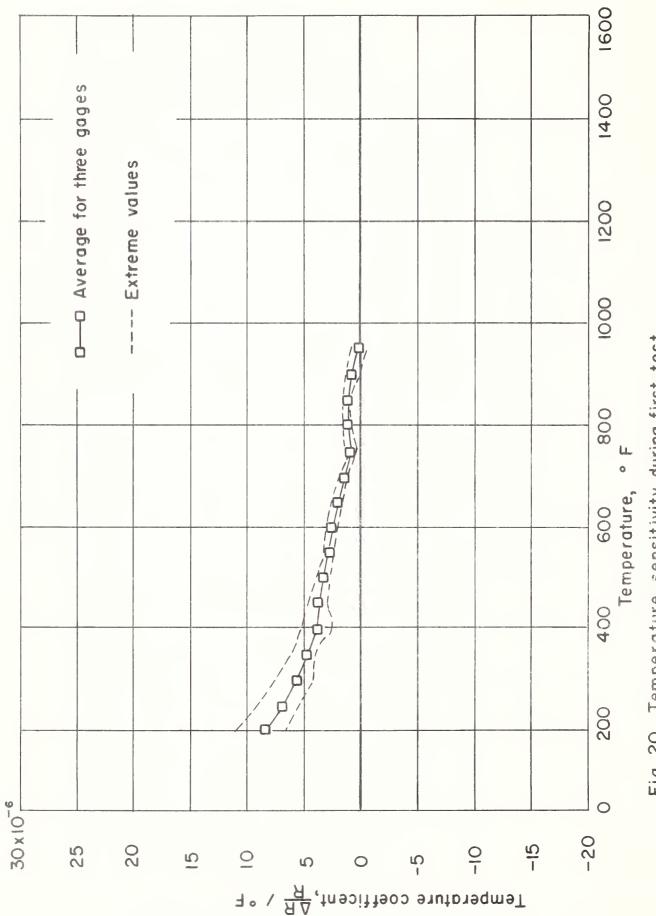


Fig. 20 Temperature sensitivity during first test.

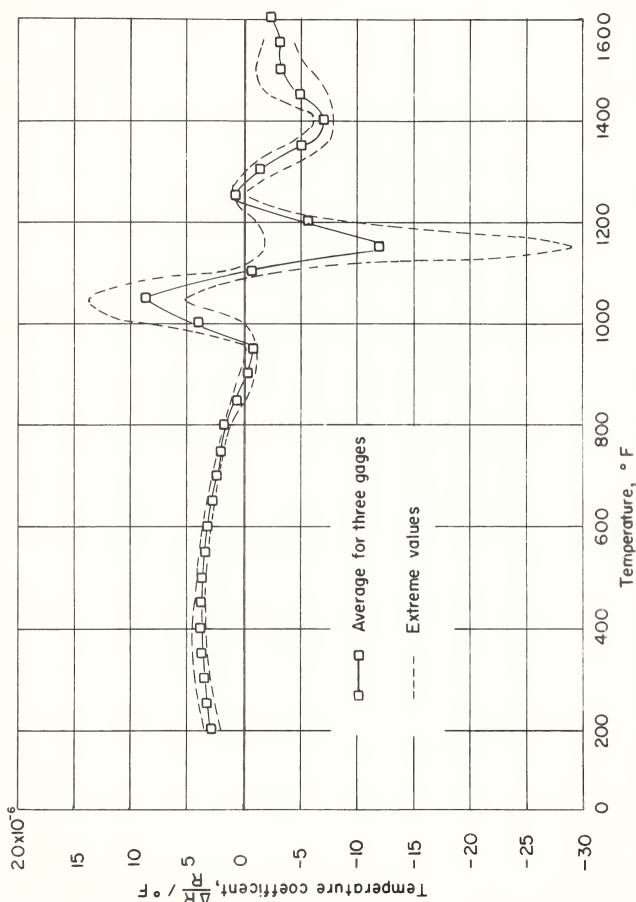


Fig. 21 Temperature sensitivity during third test

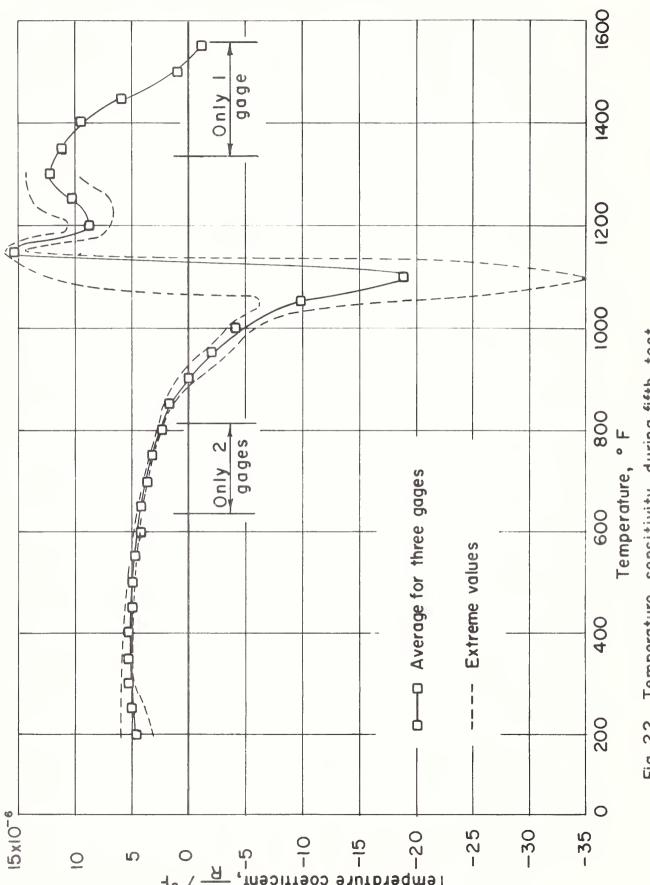


Fig. 22 Temperature sensitivity during fifth test

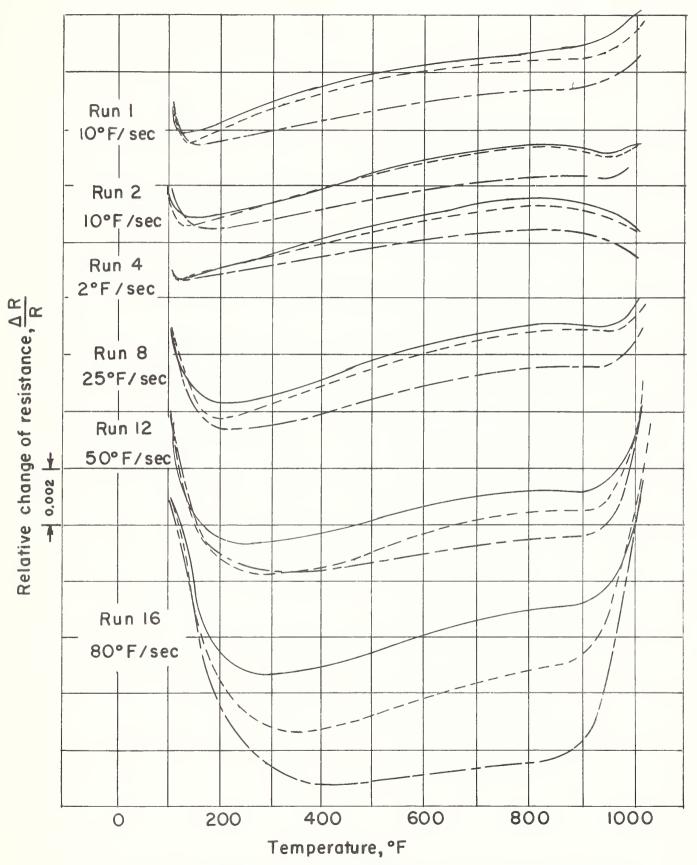


Fig. 23 Response of three gages at various heating rates

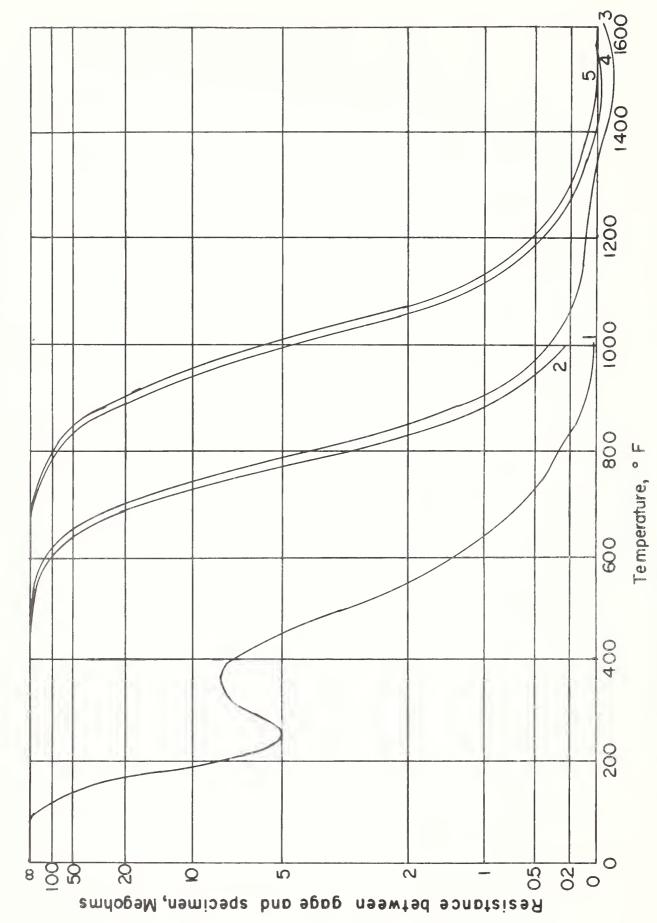


Fig. 24 Leakage resistance

## DISTRIBUTION LIST

## U. S. Government Agencies

Chief, Bureau of Naval Weapons (RAAD-23) Washington, D. C. 20360	4	Commanding Officer Air Force Flight Test Center Edwards Air Force Base, Calif. Attn: FTOTL	1
Commander, Air Force Flight Dynamics Laboratory (FDTE) Wright-Patterson Air Force Base Ohio 45433	6	U. S. Atomic Energy Commission Technical Information Service P. O. Box 62 Oak Ridge, Tennessee	1
Commander, Air Force Flight Dynamics Laboratory (ASRMDS-2) Wright-Patterson Air Force Base Ohio 45433	1	U. S. Department of Agriculture Madison Branch Forest Products Laboratory Madison 5, Wisconsin	1
Commander, Air Force Flight  Dynamics Laboratory (ASRCEM-1)  Wright-Patterson Air Force Base Ohio 45433	1	Chief, Bureau of Ships (Code 548) Washington, D. C. 20360	3
Scientific and Technical Information Facility Attn: NASA Representative (S-AK/DL)	5	Commanding Officer, Naval Air Experimental Station (ASL) Philadelphia, Pennsylvania 19112 Attn: Mr. R. Friedman	2
P. O. Box 5700 Bethesda, Maryland		Office of Naval Research (Mechanics Branch Code 438) Washington, D. C.	2
Director, National Aeronautics and Space Administration Langley Research Center Langley Field, Virginia Attn: Mr. J. Munick  Director, National Aeronautics	1	Naval Boiler and Turbine Laboratory Philadelphia Naval Base Philadelphia 12, Pennsylvania Attn: Mr. Murdock, Instrumenta- tion Division	1
and Space Administration Lewis Research Center Cleveland 11, Ohio		Naval Research Laboratory Anacostia, D. C.	2
Commanding General Aberdeen Proving Ground, Maryland Attn: Technical Library	1	Oak Ridge National Laboratory Oak Ridge, Tennessee Attn: Mr. H. J. Metz, Instrument Department	1
Commander, Air Research and Development Command Andrews Air Force Base, Maryland	2		

Commanding General, U. S. Army Ordnance Missile Command Redstone Arsenal, Alabama Attn: Technical Library	3	Springfield Armory Federal Street Springfield, Massachusetts Attn: Mr. Salame	1
Commander, Defense Documentation Genter Cameron Station Alexandria, Virginia 22314 Attn: TIPCR	10	Federal Aviation Agency NAFEC Library Building 3 Atlantic City, New Jersey	1
Harry Diamond Laboratory Electromechanical Laboratory Room 1W29, Building 92 Washington, D. C. 20438	1	National Aeronautics and Space Administration Goddard Space Flight Center Greenbelt, Maryland 20771 Attn: Charles Staugaitis Gode 623	1
National Bureau of Standards Metallic Building Materials Section Washington, D. C. 20234	1		

## Other Agencies

Advanced Technology Laboratories 369 Whisman Road Mountain View, California	1	Allison Division General Motors Corp. Indianapolis 6, Indiana	1
Aerolab Development Company, Inc. 330 W. Holly Street Pasadena 3, Galifornia	1	American Instrument Company Silver Spring, Maryland	1
AIResearch Manufacturing Co. of Arizona 402 S. 36th Street	1	A. O. Smith Corporation Milwaukee 1, Wisconsin Attn: Research Library	1
Phoenix, Arizona		Armour Research Foundation Illinois Institute of Technology	1
Allegany Instruments Co. 11801 Mississippi Avenue Los Angeles 25, California	1	Chicago 16, Illinois Attn: Mr. W. Graft	
Allegany Instruments Co. 1091 Wills Mountain Cumberland, Maryland	1	Armour Research Foundation Illinois Institute of Technology Chicago 16, Illinois Attn: Mr. H. L. Rechter	1
Allied Research Associates, Inc. 43 Leon Street Boston 15, Massachusetts Attn: Mr. D. Franklin	1	Atomics International A Division of North American Aviation, Inc. P. O. Box 309 Canoga Park, California	1

Atomic Power Development Associates, Inc. 1911 First Street	1	Cessna Aircraft Company Wichita, Kansas	1
Detroit 26, Michigan Attn: Mr. F. R. Beyer		Chance Vought Aircraft, Inc. Dallas, Texas	1
Baldwin-Lima-Hamilton Corp. Electronics and Instrumentation Division 42 Fourth Avenue	1	Columbia Research Laboratories MacDade Blvd. and Bullens Lane Woodlyn, Pennsylvania	1
Waltham 54, Massachusetts		Combustion Engineering, Inc. Chattancoga Division	1
Beech Aircraft Corporation Wichita, Kansas	1	Chattanooga 1, Tennessee	1
Bell Aircraft Corporation Niagara Falls, New York	1	Consolidated Electrodynamics Corporation 360 Sierra Madre Villa Pasadena 15, California	1
Bell Aircraft Corporation Fort Worth, Texas	1	Attn: Research Library	1
ARO, Inc. Tullahoma, Tennessee Attn: Mr. H. K. Matt	1	Convair, A Division of General Dynamics Corporation San Diego, California	1
Bendix Products Division - Missiles	1	Convair, A Division of General Dynamics Corporation Fort Worth, Texas	1
Bendix Aviation Corporation Mishawaka, Indiana Attn: George T. Cramer		Cook Research Laboratories 6401 Oakton Street Morton Grove, Illinois	1
B. J. Electronics P. O. Box 1679 Santa Ana, California	1	Cornell Aeronautical Laboratory, Inc. Structural Laboratory Section	1
Boeing Airplane Company Seattle, Washington	1	Buffalo 21, New York Attn: Mr. J. E. Carpenter	
Boeing Airplane Company Wichita, Kansas	1	Curtiss-Wright Corporation Test Instrumentation & Equipment Division	1
The Budd Company Instrument Division P. O. Box 245	1	Wood-Ridge, New Jersey Attn: Mr. M. Semanyshyn	
Phoenixville, Pennsylvania	1	Curtiss-Wright Corporation Propeller Division	1
Bulova Research & Development Laboratories, Inc. 62-10 Woodside Avenue Woodside 77, New York	1	Caldwell, New Jersey	

Douglas Aircraft Company, Inc. Santa Monica, California	1	General Electric Company Missile & Ordnance Systems Dept. 3198 Chestnut Street	1
Douglas Aircraft Company, Inc. El Segundo, California	1	Philadelphia 4, Pennsylvania	
Douglas Aircraft Company, Inc. Long Beach, California	1	Goodyear Aircraft Corporation Akron 15, Ohio	1
Esso Research and Engineering Co. P. O. Box 8 Linden, New Jersey Attn: Design Engineering Div.	1	Grumman Aircraft Engineering Corporation Bethpage, Long Island, New York Attn: Engineering Library Plant 5	1
Fairchild Aircraft Division Fairchild Engine & Airplane Corp. Hagerstown, Maryland	1	High Temperature Instruments Corp. 225 West Lehigh Philadelphia, Pennsylvania	1
Fairchild Engine Division Fairchild Engine & Airplane Corp. Deer Park, Long Island, New York	1	J. T. Hill Company 420 S. Pine Street San Gabriel, California	1
Fielden Instrument Division Robert Shaw-Fulton Controls Co. 2920 N. 4th Street Philadelphia 33, Pennsylvania	1	Hughes Aircraft Company 13141 Downie Place Garden Grove, California Attn: Mr. Philip O. Vulliet	1
Foster Wheeler Corporation 666 Fifth Avenue New York 19, New York	1	Lockheed Aircraft Corporation Burbank, California Attn: Mr. W. Brewer, Research Dept.	1
General Electric Company ANP Department Cincinnati 15, Ohio	1	Lockheed Aircraft Corporation Georgia Division Marietta, Georgia	1
General Electric Company General Engineering Laboratory	1	Attn: Engr. Tech. Library	
Schenectady, New York Attn: Mr, D, DeMichele		Lockheed Aircraft Corporation Missiles Systems Division Van Nuys, Galifornia	1
General Electric Company	1		
Aircraft Gas Turbine Division Cincinnati 15, Ohio		Lockheed Electronics Company Auronics & Industrial Products Div.	1
General Electric Company	1	Transducer Department	
Special Products Division 30th and Walnut Streets Philadelphia, Pennsylvania Attn: Mr. M. Bennon		6201 E. Randolph Street Los Angeles 22, California	

Lockheed Aircraft Corporation P. O. Box 551 Burbank, California Attn: C. J. Buzzetti	1	North American Aviation, Inc. Structures Engineering Dept. Inglewood, California	1
Bldg. 360, Plant B-6		North American Aviation, Inc. Columbus, Ohio	1
Lycoming Division	1		es.
AVCO Manufacturing Corporation Stratford, Connecticut Attn: Mr. R. Hohenberg		Northrop Aircraft, Inc. Hawthorne, California	1.
		Pennsylvania State College	1
Marquardt Corporation 16555 Saticoy Street	1	University Park, Pennsylvania	
Van Nuys, California Attn: Engineering Library Mr. Leslie Bermann		Polytechnic Institute of Brooklyn 99 Livingston Street Brooklyn 1, New York	1
Structures Development Lab		Attn: Mr. N. J. Hoff	
Massachusetts Institute of Technology	1	Research Librarian Portland Cement Association	1
Laboratory for Insulation Research Cambridge 39, Massachusetts		5420 Old Orchard Road Skokie, Illinois	
McDonnell Aircraft Corporation St. Louis, Missouri	1	Pratt and Whitney Aircraft Div. United Aircraft Corporation	1
Metrix, Inc.	1	East Hartford, Connecticut Attn: Mr. G. E. Beardsley, Jr.	
P. O. Box 683		needs. In o to be bedred tey, or o	
Walnut Creek, California		Radiation Incorporated Instrumentation Division	1
Mr. Given A. Brewer, Pres.	1	P. O. Box 2040, Fine Castle	
Brewer Engineering Laboratories		Branch	
P. O. Box 288 Marion, Massachusetts		Orlando, Florida Attn: Mr. U. R. Barnett	
mar rong massachusetts		riceis. Into the Res Barriet	
Microdot, Inc.	1	Republic Aviation Company	1
220 Pasadena Avenue South Pasadena, California		Farmingdale, Long Island, New York	
Mithra Engineering Company	1	Research, Incorporated	1
P. O. Box 472 Van Nuys, California		P. O. Box 6164 Edina Branch Post Office	
National Electronics	1	Minneapolis 24, Minnesota Attn: Mr. K. G. Anderson	
Laboratories, Inc.	T	ALLH. FIL. R. G. AHOELSOH	
1713 Kalorama Road, N. W. Washington, D. C. 20009		Ryan Aeronautical Company San Diego, California	1

Solar Aircraft Company 2200 Pacific Highway San Diego 12, California	1	Westinghouse Electric Corp. Atomic Power Division Pittsburgh, Pennsylvania	1
Southwest Research Institute 8500 Culebra Road San Antonio 6, Texas	1	Professor H. H. Bleich (NR 064-417) Dept. of Civil Engineering Columbia University	1
Statham Laboratories, Inc. 12401 W. Olympic Boulevard Los Angeles 64, California	1	Broadway at 117 Street New York 27, New York	
Stratos Division Fairchild Engine & Airplane Corp. Bay Shore, Long Island, New York	1	Professor D. C. Drucker (NR 064-424) Division of Engineering Brown University Providence 12, Rhode Island	1
Temco Aircraft Corporation Dallas, Texas	1	Professor N. J. Hoff (NR 064-425)	1
The Martin Company Baltimore, Maryland	1	Division of Aeronautical Engineering Stanford University	
The Martin Company Denver 1, Colorado	1	Stanford, California	1
Thiokol Chemical Corporation Utah Division Brigham City, Utah Attn: Instrumentation Engineer- ing Unit	1	Professor Joseph Kempner (NR 064-433)  Dept. of Aeronautical Engineering and Applied Mechanics  Polytechnic Institute of Brooklyn 333 Jay Street  Brooklyn 1, New York	1
Trans-Sonics, Inc. P. O. Box 328 Lexington 73, Massachusetts	1	Mr. Peter Stein 5602 E. Monte Rosa Phoenix, Arizona	1
University of Colorado Boulder, Colorado Attn: Prof. F. C. Walz	1	Bristol Aircraft Limited Electronic and Vibration Laboratory - E. D. L.	1
University of Dayton Research Institute Special Projects Division	1	Filton House Bristol, England	
Dayton 9, Ohio Attn: E. A. Young		Westinghouse Electric Corporation Materials Engineering Department K-70, Performance Laboratory	1
University of New Mexico Engineering Experiment Station Albuquerque, New Mexico	1	East Pittsburgh, Pennsylvania	

Westinghouse Electric Corp. Eng Mech Sect, R & D Center Beulah Road Pittsburgh 35, Pennsylvania

1



